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 748-1685

FAA Airworthiness Directives Compliance Record

Company: Michael Mayo
 Category: Airframe
 Manufacturer: Piper Aircraft, Inc.
 Model: PA-28-140

Position:
 P/N:
 S/N: 28-25633

Aircraft Registration No: N8792N
 ATP Revision: 02/15/2020

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2018-07-03 04/20/2018	To allow the pilot to do a preflight check of the fuel selector placards	02/17/2020 Hrs: -- C: --	CW by visual inspection - placards are properly located.	No	D: -- Hrs: -- C: --	
2017-14-04 08/15/2017	[Recurring] To prevent rupture or failure of the oil cooler hose assemblies, which could result in engine stoppage with,contd.	02/17/2020 Hrs: 4029.47 C: --	CW by visual inspection of the firesleeve - no defects noted. Next inspection due at 4129.47 hours tach time. New hoses installed 07/27/2017, replacement due at 4505.76 hours tach time or 06/2025, whichever occurs first.	Yes	D: -- Hrs: 4129.47 4220.2 C: -- see log on 10/30/20 - 4120.2	John V. White A&P 303543 IA <i>[Signature]</i>
2013-02-13 03/11/2013	[Recurring] To prevent failures that may lead to failure of the horizontal stabilator control system and cold result,contd.	02/15/2014 Hrs: 3494.6 C: --	PCW by visual inspection, 01/15/14. Next due at 5494.6 hours tach time or 02/13/2021, whichever occurs first.	Yes	D: 02/13/2021 2/13/21 Hrs: 5494.6 5494.6 C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2010-15-10 08/31/2010	To detect and correct any incorrectly assembled control wheel shafts	12/11/2011 Hrs: -- C: --	PCW by visual inspection - no defects found, 12/11/2011.	No	D: -- Hrs: -- C: --	✓
99-05-09 03/19/1999	To prevent pieces of a damaged induction air filter from being ingested into the engine, which could, contd.	04/20/1999 Hrs: -- C: --	CW by installation of Brackett Air Filter BA-104, 04/20/1999.	No	D: -- Hrs: -- C: --	✓
96-10-01 R1 12/14/1998	TO PREVENT THE LANDING LIGHT SEAL FROM LODGING IN THE CARBURETOR, WHICH, IF NOT DETECTED & CORRECTED, CONTD.	07/07/2000 Hrs: -- C: --	PCW by replacement of landing light, 07/07/2000.	No	D: -- Hrs: -- C: --	✓
96-10-03 06/14/1996	TO PREVENT FAILURE OF THE FLAP HANDLE ATTACH BOLT & SUDDEN RETRACTION OF THE FLAPS, WHICH, IF NOT DETECTED, CONTD.	02/26/1998 Hrs: -- C: --	PCW, 02/26/1998.	No	D: -- Hrs: -- C: --	✓
69-22-02 R(2) 07/30/1979	[Recurring] TO PREVENT POSSIBLE FAILURE OF THE CONTROL WHEEL	02/17/2020 Hrs: 4029.47 C: --	CW by visual inspection - no defects noted.	Yes	D: -- Hrs: 4129.47 4220.2 C: -- see log @ 4120.2	✓
77-23-03 11/14/1977	TO PREVENT A POWER LOSS DUE TO THERMAL CONTRACTION BINDING A CONTROL ROD END	01/22/1980 Hrs: -- C: --	PCW by visual inspection, 01/22/1980. Correct rod ends installed.	No	D: -- Hrs: -- C: --	✓
77-01-01 01/07/1977	TO PREVENT POSSIBLE FUEL EXHAUSTION DUE TO AN ERRONEOUS FUEL QUANTITY GAUGE INDICATION	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	✓
75-08-03 04/04/1975	TO DETECT AND CORRECT UNSECURED FEL DRAIN VALVES INSTALLED AT THE GASCOLATOR	09/11/1975 Hrs: -- C: --	PCW, 09/11/1975.	No	D: -- Hrs: -- C: --	✓
75-02-03 11/26/1974	TO PREVENT POSSIBLE FAILURE OF THE NOSE WHEEL FORK ASSEMBLY	-- Hrs: -- C: --	NA - Affected STC not installed.	No	D: -- Hrs: -- C: --	✓

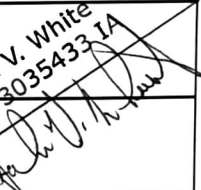
John V. White
A&P 3035433 IA
[Signature]

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
74-13-04 06/21/1974	TO PREVENT THE LOSS OF POWER DUE TO A FAILED BALL JOINT ASSEMBLY	09/11/1975 Hrs: -- C: --	PCW by installation of Piper Kit No. 760- 780V on 09/11/1975.	No	D: -- Hrs: -- C: --	
74-09-04 04/24/1974	TO PROVIDE A SAFER REAR SEAT BELT INSTALLATION	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	
74-06-02 03/18/1974	[Recurring] TO PREVENT POSSIBLE LEAKAGE OF CARBON MONOXIDE INTO THE CABIN HEATER SYSTEM	-- Hrs: -- C: --	NA - Affected STCs not installed.	No	D: -- Hrs: -- C: --	
72-24-02 11/17/1972	TO PREVENT POWER INTERRUPTION AS A RESULT OF ABRUPT THROTTLE MOVEMENT	06/06/1973 Hrs: -- C: --	PCW by installation of required placard, 06/06/1973.	No	D: -- Hrs: -- C: --	
72-14-07 07/07/1972	TO INSURE PROPERLY TIGHTENED BOLTS ON THE STABILATOR HINGE ATTACHMENT FITTINGS, WHICH ARE SECURED TO THE AFT, CONTD.	06/06/1973 Hrs: -- C: --	PCW, 06/06/1973.	No	D: -- Hrs: -- C: --	
70-16-05 05/19/1972	[Recurring] TO PREVENT FAILURE OF THE ENGINE EXHAUST MUFFLER	02/17/2020 Hrs: 4029.47 C: --	CW by visual inspection - no defects noted. New muffler was installed at 3687.59 hours. Per para (b) inspection interval is 100 hours until 950 hrs TIS is reached, then every 50 hrs. Next due at 4129.47 hours tach time.	Yes No	D: -- Hrs: 4129.47 C: -- Non recurring, see log book 10/30/20 - 4120.2	
72-08-06 04/14/1972	[Recurring] TO DETECT CRACKS ADJACENT TO THE 2 1/2" DIAMETER MACHINED BOSS IN ANY OF THE FOUR MAIN LANDING GEAR TORQUE LINKS	-- Hrs: -- C: --	NA - Affected torque links not installed. P/Ns 78031-0 are installed on aircraft.	No	D: -- Hrs: -- C: --	

John V. White
A&P 3035433 IA
[Signature]

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
71-21-08 10/13/1971	TO PREVENT POSSIBLE BINDING OF THE FUEL SELECTOR HANDLE	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	
71-14-06 07/09/1971	TO PREVENT POSSIBLE INADVERTENT GROUNDING OF MAGNETOS	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	
70-26-04 R(2) 06/24/1971	[Recurring] TO DETECT CRACKS IN THE STABILATOR BALANCE WEIGHT TUBE IN THE AREA OF THE ATTACHMENT BOLT HOLES	05/18/2018 Hrs: 3887.88 C: --	CW by dye penetrant inspection. Next due at 4087.88 hours tach time.	Yes No	D: -- Hrs: 4087.88 C: --	No recurring see log book on 8/24/20 - 4088.5
70-18-05 09/07/1970	TO PRECLUDE THE POSSIBILITY OF FAILURE OF THE MAIN LANDING GEAR TORQUE LINK BOLTS	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	
67-26-02 05/22/1968	VARIOUS MODIFICATONS	-- Hrs: -- C: --	NA by aircraft model.	No	D: -- Hrs: -- C: --	
67-26-03 04/12/1968	[Recurring] TO PREVENT THE POSSIBILITY OF RESTRICTING FUEL FLOW THROUGH THE FUEL SYSTEM SCREENS AND FILTERS CAUSED BY, CONTD.	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	
67-20-04 09/27/1967	TO PRECLUDE THE POSSIBILITY OF FAILURE OF THE MAIN LANDING GEAR TORQUE LINK	-- Hrs: -- C: --	NA by date of manufacture.	No	D: -- Hrs: -- C: --	
67-12-06 01/01/1967	DUE TO THE POSSIBILITY OF INTERNAL CORROSION RESULTING INADEQUATE CORROSION PROTECTION OF CERTAIN, CONTD.	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	
65-06-06 03/12/1965	[Recurring] TO PREVENT FURTHER FAILURES OF THE FUEL QUANTITY GAUGE SENDER UNIT TERMINAL CONNECTING WIRE THAT COULD CAUSE, CONTD.	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	

John V. White
A&P 3035433 IA
[Signature]

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64-06-06 04/06/1964	[Recurring] TO PRECLUDE FAILURE OF THE CONTROL WHEEL	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	 John V. White A&P 3035433 1A 
62-19-03 08/28/1962	TO PRECLUDE THE LOSS OF THE PROPELLER DUE TO THE FAILURE OF THE BOLTS	-- Hrs: -- C: --	NA by aircraft S/N.	No	D: -- Hrs: -- C: --	

Category: Engine

Position:

ATP Revision: 02/15/2020

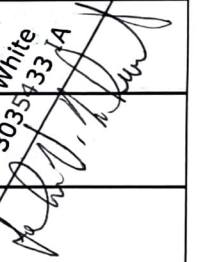
Manufacturer: Lycoming

P/N:

Model: O-320-E2A

S/N: L-25051-27A

original

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2017-16-11 08/15/2017	To prevent connecting rod failure which could result in uncontained engine failure, total engine power loss, contd.	-- Hrs: -- C: --	NA - No engine work requiring the replacement of connecting rods was performed during the affected time frame.	No	D: -- Hrs: -- C: --	 John V. White A&P 3035433 1A 
2015-02-07 03/11/2015	To prevent the propeller governor shaft set screw from coming loose, causing damage to the engine and, contd.	-- Hrs: -- C: --	NA - Engine is not aerobatic and not equipped with a propeller governor.	No	D: -- Hrs: -- C: --	
2012-03-07 03/27/2012	To prevent engine in-flight shutdown, power loss, and reduced control of the airplane	-- Hrs: -- C: --	NA - Engine is not equipped with an HA -6 carburetor.	No	D: -- Hrs: -- C: --	
2009-26-12 02/04/2010	[Recurring] To prevent loss of engine power due to cracks at the head-to-barrel interface and possible engine failure, contd.	03/31/2012 Hrs: -- C: --	NA by cylinder assembly P/Ns, see log entry dated 03/31/2012.	No	D: -- Hrs: -- C: --	
2007-04-19 R1 05/07/2007	To prevent cylinder separation that can lead to engine failure, possible engine compartment fire, and, contd.	-- Hrs: -- C: --	NA - Engine not equipped with Superior cylinders.	No	D: -- Hrs: -- C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2006-12-07 07/11/2006	To prevent loss of engine power due to cracks in the cylinder assemblies & possible engine failure caused,contd.	03/31/2012 Hrs: -- C: --	NA by cylinder assembly P/Ns, see log entry dated 03/31/12.	No	D: -- Hrs: -- C: --	
2004-10-14 C 06/25/2004	[Recurring] To prevent loosening or failure of the crankshaft gear retaining bolt, which may cause sudden engine failure	04/06/1990 Hrs: -- C: --	PCW during OH, 04/06/1990. Due at next engine OH or in the event of prop strike/sudden stoppage.	Yes	D: -- Hrs: -- C: --	
98-17-11 C 10/19/1998	TO PREVENT CRANKSHAFT FAILURE DUE TO CRACKING, WHICH COULD RESULT IN AN INFLIGHT ENGINE FAILURE AND POSSIBLE,CONTD.	-- Hrs: -- C: --	NA - Crankshaft not serviced by this vendor.	No	D: -- Hrs: -- C: --	
98-02-08 03/30/1998	[Recurring] TO PREVENT CRANKSHAFT FAILURE, WHICH CAN RESULT IN ENGINE FAILURE, PROPELLER SEPARATION, FORCED LANDING, AND,CONTD.	05/18/2018 Hrs: -- C: --	CW by visual inspection - no defects noted. Due at next engine OH, disassembly or by 05/17/2023, whichever occurs first.	Yes	D: 05/17/2023 Hrs: -- C: --	
97-15-11 08/12/1997	TO PREVENT PISTON PIN FAILURE, WHICH COULD RESULT IN ENGINE FAILURE	-- Hrs: -- C: --	NA - Lycoming cylinder assemblies or piston pins were not installed within the affected time frame.	No	D: -- Hrs: -- C: --	
96-09-10 C 07/15/1996	TO PREVENT OIL PUMP FAILURE DUE TO IMPELLER FAILURE, WHICH COULD RESULT IN AN ENGINE FAILURE	08/30/1981 Hrs: -- C: --	NA - Per para (c)(2), no action required if AD 81-18-04 was PCW. See log entry dated 08/30/1981.	No	D: -- Hrs: -- C: --	
95-26-02 01/24/1996	TO PREVENT DETONATION DUE TO LOW OCTANE, WHICH CAN RESULT IN SEVERE ENGINE DAMAGE AND SUBSEQUENT FAILURE	-- Hrs: -- C: --	NA by aircraft registration number.	No	D: -- Hrs: -- C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
92-12-05 07/10/1992	TO PREVENT PISTON PIN FAILURE, OR PISTON RELEASE, AND ENGINE FAILURE	-- Hrs: -- C: --	NA - Cylinders not removed during affected time frame. LW-14077 piston pins not installed.	No	D: -- Hrs: -- C: --	
90-04-06 R1 05/28/1991	TO PREVENT OIL LINE FRACTURE AND LOSS OF ENGINE OIL	-- Hrs: -- C: --	NA - Engine not equipped with a propeller governor.	No	D: -- Hrs: -- C: --	
87-10-06 R1 09/01/1989	TO PREVENT POSSIBLE ROCKER ARM FAILURE AND LOSS OF ENGINE POWER, INSPECT AND REWORK OR REPLACE ROCKER ARM, CONTD.	-- Hrs: -- C: --	NA - Engine not OH'd or had rocker arms replaced during affected time frame.	No	D: -- Hrs: -- C: --	
75-08-09 R(3) 08/18/1977	TO PREVENT OIL PUMP FAILURES, INSPECT, REPLACE AND ASSEMBLE THE OIL PUMP DRIVE SHAFT AND DRIVE IMPELLER	05/12/1979 Hrs: -- C: --	PCW during OH, 05/12/1979	No	D: -- Hrs: -- C: --	
63-23-02 08/06/1968	[Recurring] EXHAUST VALVES AND VALVE GUIDES	-- Hrs: -- C: --	NA by date of manufacture.	No	D: -- Hrs: -- C: --	
66-20-04 08/27/1966	TO PREVENT FURTHER FAILURES OF OIL FILTER ADAPTER GASKET, P/N 74904	-- Hrs: -- C: --	NA by date of manufacture.	No	D: -- Hrs: -- C: --	
64-16-05 07/10/1964	TO PRECLUDE THE POSSIBILITY OF ENGINE OIL BEING DRAINED OVERBOARD AS A RESULT OF OIL SEAL FAILURE, CONTD.	-- Hrs: -- C: --	NA by date of manufacture.	No	D: -- Hrs: -- C: --	
59-10-07 07/01/1959	CYLINDER BAFFLE CLAMPS	-- Hrs: -- C: --	NA by date of manufacture.	No	D: -- Hrs: -- C: --	
55-02-02 02/28/1955	ENGINE ACCESSORY DRIVE ADAPTER GASKET	-- Hrs: -- C: --	NA by date of manufacture.	No	D: -- Hrs: -- C: --	

Category: Propeller

Position:

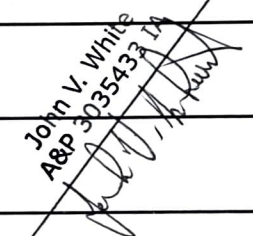
ATP Revision: 02/15/2020

Manufacturer: Sensenich Corp.

P/N:

Model: 74DM6

S/N: K-28105

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-14-11 08/17/2005	To prevent blade failure that could result in separation of a propeller blade and loss of control of the airplane	-- Hrs: -- C: --	NA - Propeller not serviced by this vendor.	No	D: -- Hrs: -- C: --	
2003-13-17 07/18/2003	To detect unsafe conditions that could result in separation of a propeller blade & loss of control, contd.	-- Hrs: -- C: --	NA - Propeller not serviced by this vendor.	No	D: -- Hrs: -- C: --	
66-23-02 05/16/1967	TO PREVENT FURTHER MID-BLADE PROPELLER FAILURES RESULTING FROM CORROSION PITS, CONTD.	-- Hrs: -- C: --	PCW - Propeller blades do not have decals.	No	D: -- Hrs: -- C: --	
60-11-08 11/14/1961	[Recurring] CRACKED HUBS	-- Hrs: -- C: --	NA - Propeller installed on an "E" series engine and S/N prefix is a "K".	No	D: -- Hrs: -- C: --	

Category: Carburetors

Position:

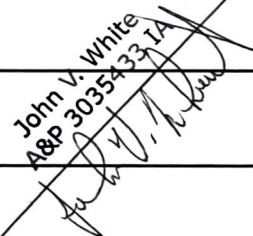
ATP Revision: 02/15/2020

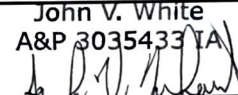
Manufacturer: Marvel-Schebler

P/N: 10-3678-32

Model: MA-4SPA

S/N: MS00XJ12

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
98-01-06 02/13/1998	[Recurring] TO PREVENT DISRUPTION OF FUEL FLOW TO THE ENGINE RESULTING IN FAILURE TO ATTAIN RATED POWER, POWER LOSS IN, CONTD.	02/28/2018 Hrs: -- C: --	PCW at carb OH, 02/28/2018. Dataplate stamped with "V".	No	D: -- Hrs: -- C: --	
93-19-04 C 10/18/1993	TO PREVENT A DISRUPTION OF FUEL FLOW TO THE ENGINE, RESULTING IN ENGINE POWER LOSS, ENGINE FAILURE, CONTD.	02/28/2018 Hrs: -- C: --	PCW at carb OH, 02/28/2018	No	D: -- Hrs: -- C: --	
72-06-05 R2 07/03/1986	TO PREVENT LOOSENESS OR SEPARATION OF THE THROTTLE ARM	02/28/2018 Hrs: -- C: --	PCW at carb OH, 02/28/2018.	No	D: -- Hrs: -- C: --	

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
69-24-03 11/29/1969	TO PREVENT POSSIBLE POWER LOSSES DUE TO BLOCKAGE OF CARBURETOR METERING PASSAGES BY THREAD LUBRICANT, CONTD.	02/28/2018 Hrs: -- C: --	PCW at carb OH, 02/28/2018	No	D: -- Hrs: -- C: --	John V. White A&P 3035433 IA 

Category: Magnetos

Position: LEFT

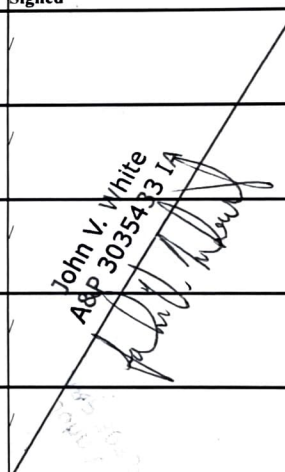
ATP Revision: 02/15/2020

Manufacturer: Bendix Corporation

P/N: 10-51360-45

Model: S-20 SERIES

S/N: 996374

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2005-12-06 07/19/2005	[Recurring] To prevent failure of the magneto impulse coupling assembly and possible engine failure	-- Hrs: -- C: --	NA - Magneto not installed on Lycoming O-540 series engine.	No	D: -- Hrs: -- C: --	
94-01-03 R2 06/28/1995	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE	07/07/2000 Hrs: -- C: --	PCW, 07/07/2000.	No	D: -- Hrs: -- C: --	
94-06-09 05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO	-- Hrs: -- C: --	NA by magneto model & P/N.	No	D: -- Hrs: -- C: --	
82-20-01 06/14/1983	TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE, CONTD.	02/24/1983 Hrs: -- C: --	PCW IAW S.B. #623, 02/24/1983. Dataplate stamped with "F".	No	D: -- Hrs: -- C: --	
74-26-09 12/24/1974	S-20,-200,-1200 SERIES MAGNETOS	09/11/1975 Hrs: -- C: --	PCW by replacement of bushing, 09/11/1975.	No	D: -- Hrs: -- C: --	

Category: Magnetos

Position: RIGHT

ATP Revision: 02/15/2020

Manufacturer: Bendix Corporation

P/N: 10-163045-3

Model: S-200 SERIES

S/N: 999046

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
94-01-03 R2 06/28/1995	TO PREVENT MAGNETO FAILURE AND SUBSEQUENT ENGINE FAILURE	05/04/2010 Hrs: -- C: --	PCW by replacement of coil & rotor, 05/04/2010.	No	D: -- Hrs: -- C: --	
94-06-09 05/20/1994	TO PREVENT INJURY OR DEATH TO GROUND PERSONNEL DUE TO A NON- GROUNDED MAGNETO	-- Hrs: -- C: --	NA by magneto model & P/N.	No	D: -- Hrs: -- C: --	
82-20-01 06/14/1983	TO PREVENT FAILURE OF IMPULSE COUPLING DUE TO IMPROPERLY HEAT TREATED (SOFT) FLYWEIGHTS RESULTING IN ENGINE,CONTD.	-- Hrs: -- C: --	NA by magneto P/N.	No	D: -- Hrs: -- C: --	
74-26-09 12/24/1974	S-20,-200,-1200 SERIES MAGNETOS	-- Hrs: -- C: --	NA by magneto P/N.	No	D: -- Hrs: -- C: --	

Category: Alternators

Manufacturer: Falcon

Model: 2642997

Position:

P/N:

S/N:

ATP Revision: 02/15/2020

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
-- NO A&P'S		-- Hrs: -- C: --		--	D: -- Hrs: -- C: --	N/A

Category: Vacuum System

Manufacturer: RAPCO

Model: RAP215CC

Position:

P/N:

S/N:

ATP Revision: 02/15/2020

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Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
NOAD'S		-- Hrs: -- C: --		--	D: -- Hrs: -- C: --	N/A

Category: Global Positioning System

Position:

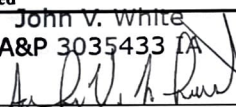
ATP Revision: 02/15/2020

Manufacturer: Garmin International

P/N:

Model: GNS 430

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
2001-23-17 12/28/2001	To prevent external noise from causing inaccurate course deviation displays in the GNS 430 unit's course, contd.	-- Hrs: -- C: --	NA by P/N.	No	D: -- Hrs: -- C: --	John V. White A&P 3035433 IA 

Category: ATC Transponder/ System

Position:

ATP Revision: 02/15/2020

Manufacturer: Garmin International

P/N:

Model: GTX 345

S/N:

Issue Number Effective Date Amendment #	Description	Complied	Method of Compliance	Recur	Next Due	Facility Cert No. / Type Authorized By Signed
NOAD'S		-- Hrs: -- C: --		--	D: -- Hrs: -- C: --	N/A